

IEEE Aerospace Conference 2019

Regolith Particle Erosion of Material in Aerospace Environments

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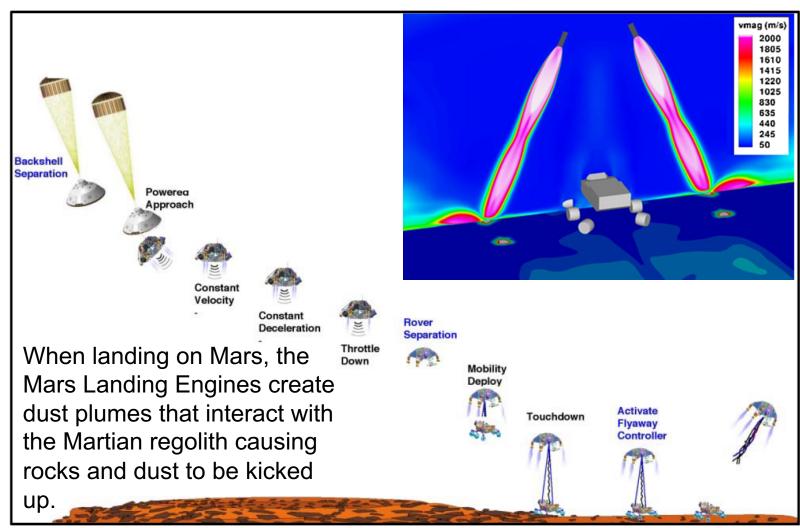
Outline

- Background
 - Extreme landing environment, which put the rover and the material at risk
- Material Selection
- Test Facility
 - Test matrix was created based on capabilities and requirements

- Results
 - Paint
 - Flex Cable
 - Composite
 - Fiber Optic Cable
- Conclusion

Background

Entry Descent and Landing (EDL)



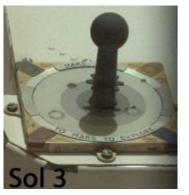
Images courtesy of:

Sengupta, Anita, J. Kulleck, J. Van Norman, and M. Mehta. "Thermal Coating Erosion in a Simulated Martian Landing Environment." Wear, vol. 270, no. 5-6, 2011, pp. 335–343., doi:10.1016/j.wear.2010.09.013.

M. Schoenenberger, A. Dyakonov, P. Buning, W. Scallion, and J. Van Norman. Aerodynamic challenges for the Mars Science Laboratory entry descent and landing, in: 41st AIAA Thermophysics Conference, 22–25 June, San Antonio, Texas, 2009.

Particle Migration on the Rover Deck









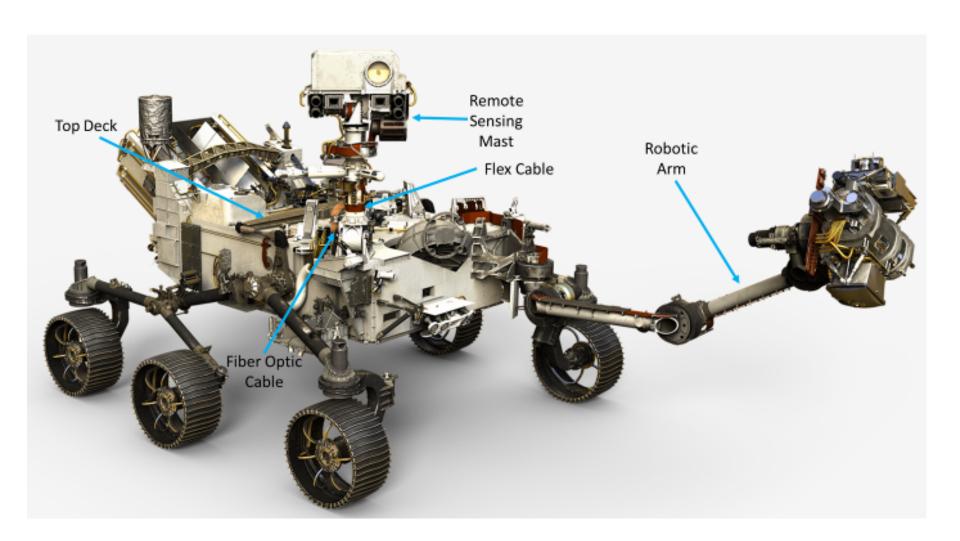
Mars Science Laboratory (MSL) immediately after landing (left) and throughout the mission life (right).

Materials Tested

- Materials that were considered at risk
 - Exposure/location
 - Durability
 - Single point failure
- Tested to determine survivability

- Thermal Control Paints
 - S13GP:6N/LO-I (S13)
 - Primers: 4044 and 4155
 - Thinners: X-99 and Xylene
 - Aptek 2711
 - MSL paint
- Flex Cable
- M55J composite
- HEPA Filter
- Fiber Optic Cable

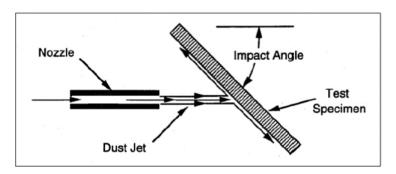
Mars 2020 Rover (M2020)



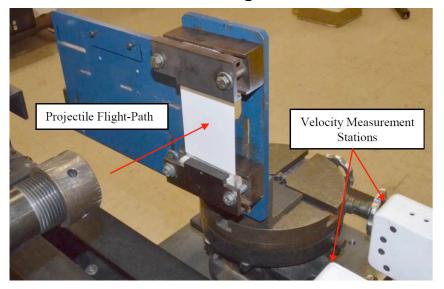
Test Setup

University of Dayton Research Institute (UDRI)

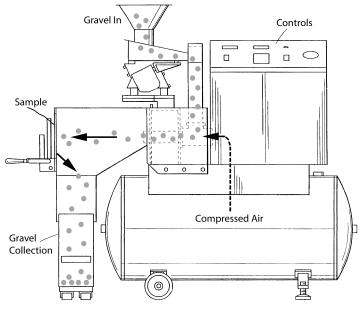
Capabilities



Erosion Rig



Gas Gun



Gravelometer

All instruments used pressurized air to accelerate the particles and lasers to measure particle velocity.

University of Dayton Research Institute (UDRI)

Erosion Media

Test Apparatus	Erosion Rig		Gravelometer	Gas Gun	
Nomenclature	SAND		GRAVEL	ROCK	
Material	Quartz		Basalt	Tonalite	
Test Media	Foundry Sand	Golf Sand	Mojave Mars Simulant (MMS) [13]	Cedar City Tonalite	
Particle Size	38-44 μm	170-550 μm	1-10 mm	6.4-15.8 mm	
Color	or brown I		red/dark gray	brown	
Shape	round	sharp and angular	irregular	machined to a 118° point	



Gravel



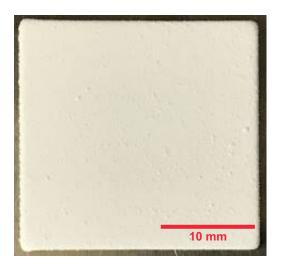
Rock

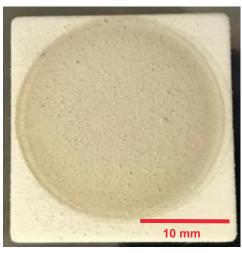
Test Matrix

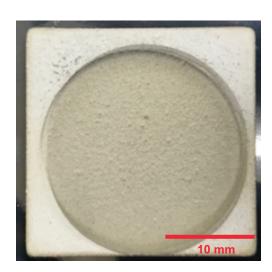
	Test Apparatus:	Erosion Rig						Gravelometer	Gas Gun						
	Particle Size:	38-44 μm		177-250 μm		240-550 μm		1-10 mm	9.5 mm	12.7 mm	15.8 mm				
	Velocity:		250 m/s	S	160 m/s	80	m/s	85	m/s	25 m/s	20 m/s	22 m/s	22 m/s	25 r	n/s
	Impact Angle:	30°	60°	90°	30°	30°	90°	30°	90°	90°	90°	90°	90°	30°	90°
	S13 Paint	X	X	X	x	X	X	X	X	X				X	X
	Aptek 2711 Paint				x										
M A T	Xylene Paint				x					X					
E R I A	Flex Cable						X	x	X	X					X
L	M55J Composite						X	X	X	X					x
	HEPA Filter						X	x	x	X					
	Fiber Optic Cable						X		X		X	Х	x		

Results

Discoloration of S13 Paint







Left: Pristine S13 coupon

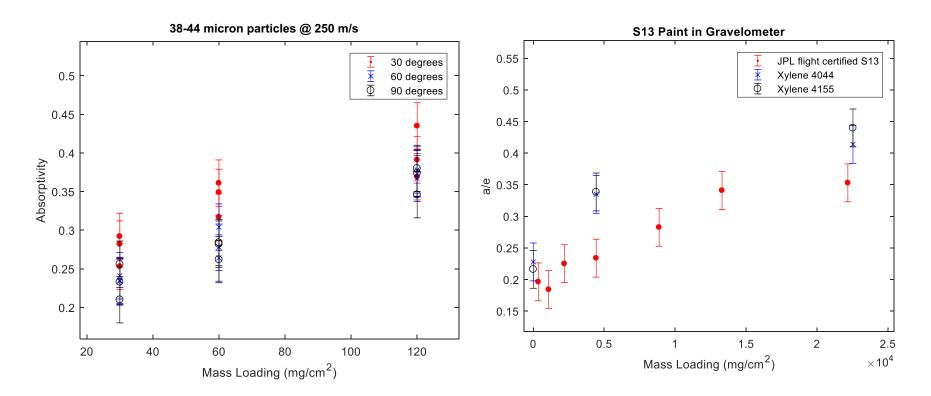
Center: S13 coupon exposed to 38-44 micron particles traveling at 225 m/s, 120

mg/cm² of total exposure

Right: The same sample after 2000 mg/cm² exposure.

Effect: The small particles embed themselves into the soft silicone paint and creating a mass gain and an increase in absorption. Increased absorption effects rover thermal control. The sand used in this erosion test was a white/brown sand; Martian regolith is red so the actual absorption value will differ.

Optical Properties of S13



Left: Impact angle had large effect on absorptivity. Right: The thinners and primers also had an effect on absorptivity. Emissivity stayed constant.

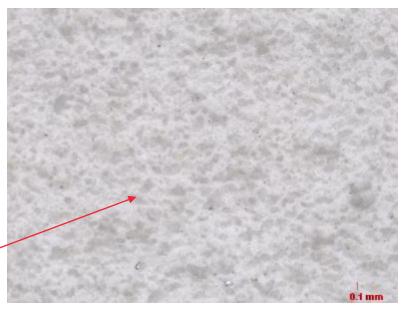
Sand Embedding in S13 Paint



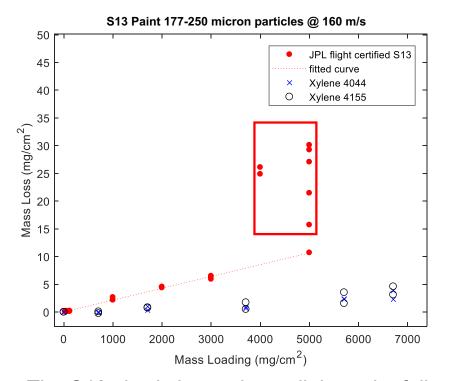
Paint before exposure.

Particles embedded in the paint after exposure to 3000mg at 160m/s 30 degrees 177-250 micron





Erosion of S13





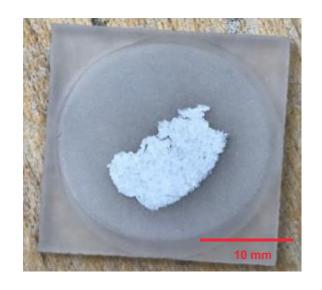
The S13 slowly lost paint until the paint fail catastrophically, detaching from the substrate.

Left: Red box shows which coupons failed.

Right: Image of paint coupon after failure.

Failure of S13 Paint

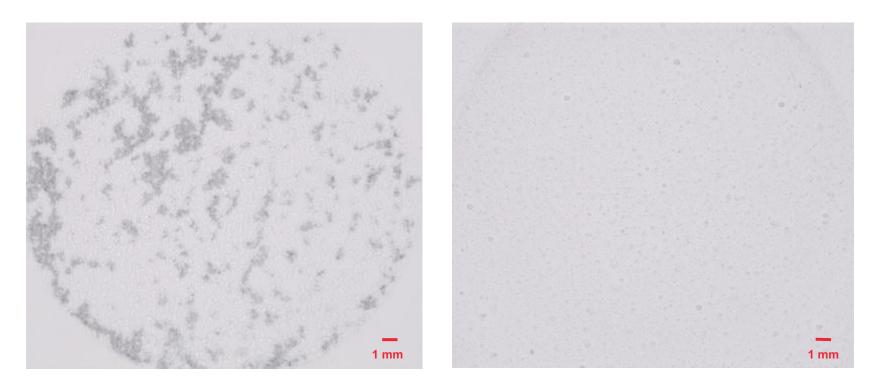




S13 paint coupon exposed to 177-250 micron particles at 160 m/s after 5 g/cm².

Right: Paint chip found in the sand after testing. The paint chip is about 0.138 in², this is 24% of the exposed surface area. This shows the failure mode of the paint is a adhesive failure to the Al substrate.

MSL vs M2020 Paint



Aptek 2711 coupon from Mars Science Laboratory (MSL) (left) and S-13 coupon (right) exposed to the same environment.

Aptek 2711 is a brittle paint and is prone to chipping.

S13 is a silicone based paint and is soft and prone to peeling but also particle embedding.

Flex Cable Erosion

Sand

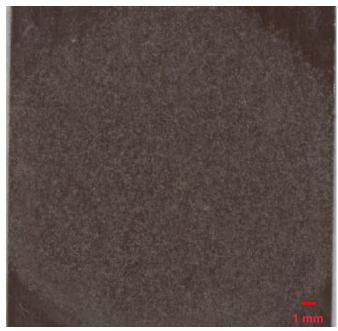
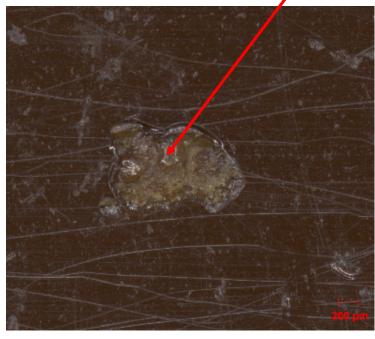


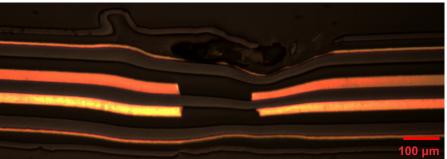


Image and cross-section shows only erosion of the Kapton outer layer.

Metal trace is not exposed

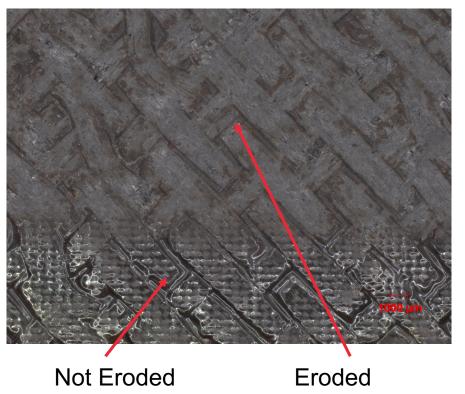
Gravel and Rock





Worst Case: Cross-section shows no evidence of shorts or opens

M55J Composite Erosion





Worst Case: Broken fibers and microscopic through holes.

M55J Composite Gas Gun Particle Impact

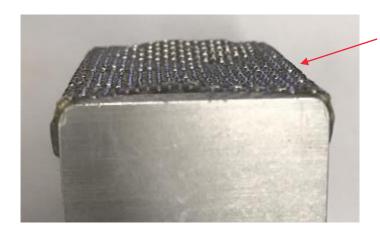




Static deformation of about 8 mm after impact from one 15.8 mm particle traveling at 25 m/s

HEPA Filter Erosion

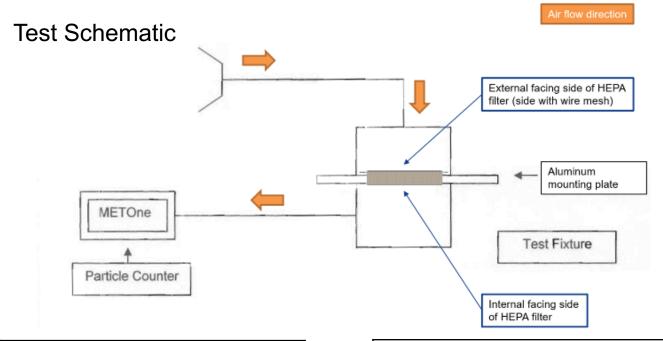




Screen deformed from particles larger than 2 mm.

No evidence of damage from particles < 2 mm.

HEPA Post-Test Particle Count



Baseline Measurement				
Time	Count			
min	0.3 micron	0.5 micron		
1	116101	16139		
2	116820	16830		
3	123836	17903		
4	121680	17596		
5	113146	16456		
6	122199	17719		
Average	118963.6667	17107.16667		
Baseline	0.3 micron	0.03% of Average		
	118963.6667	35.7		

RSM Chassis HEPA Filter Post- Test Measurement						
Time	Count					
min	0.3 micron	0.5 micron				
1	23	4				
2	14	3				
3	13	0				
4	16	0				
5	19	0				
Average	17	1.4				
Result	Pass					
Highest Value	23					

Fiber Optic Cable (FOC)

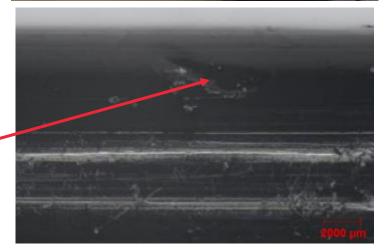
Deformed ~2"

Deformed <1.15"



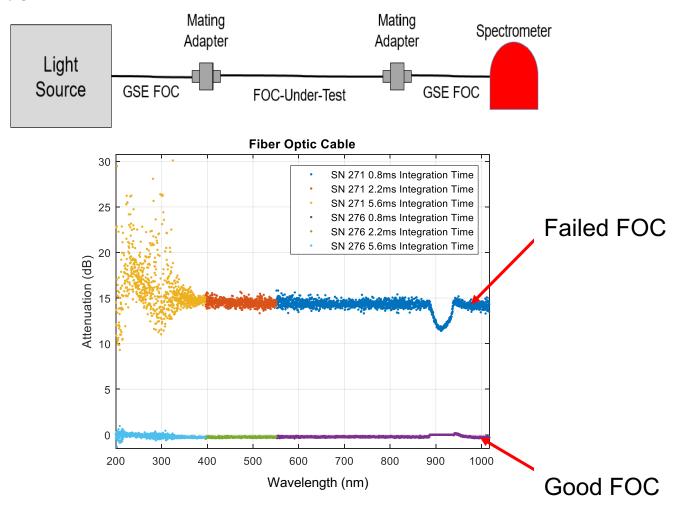


Indentation in the FOC jacket



FOC Attenuation Measurements

Test Schematic



Conclusion

- All material tested will survive the predicted high-velocity particle environment that the M2020 rover is expected to be exposed to when it lands on Mars
- An increase in the α/ε ratio due to particle embedment for the S13 paint was observed; this change in optical properties needs to be accounted for in a system level thermal analysis.
 - Once particles are embedded in the paint, there is a low likelihood of the particles migrating off of the painted surface, which means that the increased α/ε ratio will likely remain
- The paints with 4155-xylene and 4044-xylene where able to withstand higher velocities and mass loading then the current M2020 paint.
- This testing will help plan future missions to Mars and other aerospace locations.

Thank you to...

- Caltech, JPL, and NASA
- Mars 2020 Program Office
- University of Dayton Research Institute
 - Cheryl Castro
 - Kevin Poormon
 - Matt Rothgeb

Biography



Emma Bradford received a B.S. in Applied Physics with a minor in Chemistry from Xavier University in 2014. She has been at JPL for 4 years and works in the Materials Laboratory. There she tests materials in simulated space environments. Missions Emma has worked on include InSight, Mars 2020, MAIA, DSOC, and Psyche.



Jason Rabinovitch is a Mechanical Engineer at the Jet Propulsion Laboratory (JPL), California Institute of Technology, where he works in the Entry, Descent, and Landing & Formulation Group. Prior to JPL, Dr. Rabinovitch received a B.Sc. in Mechanical Engineering from Yale University in 2008, a M.Sc. in Aerospace Engineering from the California Institute of Technology in 2009, a M.Sc. in Fluid Mechanics from École Polytechnique (Paris) in 2010, and a Ph.D. in Aeronautics from Caltech in 2014. Dr Rabinovitch is a member of the AIAA Thermophysics Technical Committee, and his research interests span a wide range of topics related to experimental and computational fluid mechanics applied to EDL, vehicle design, propulsion, ablation, and geophysical applications.



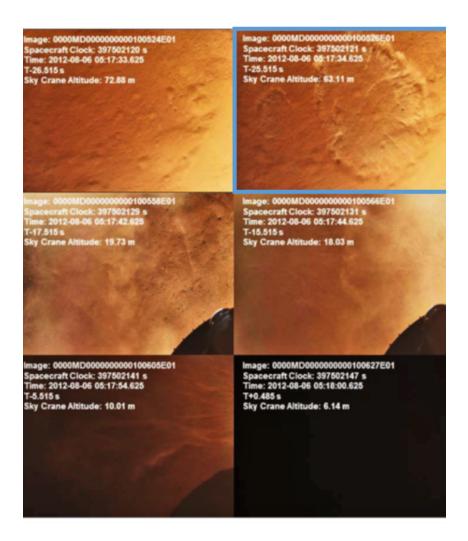
Dr. Mohamed Abid received PhD in AME from University of Southern California. He is the Deputy Chief Mechanical Engineer for M2020. He was the Manager for the Mechatronics group at JPL, and the Mission Chief Engineer of Soil Moisture Active Passive (SMAP) mission. He is the author of the textbook: "Spacecraft Sensors" a John Wiley & Sons publication.



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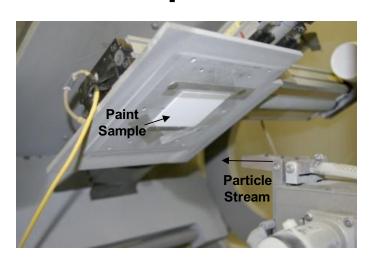
Backup Slides

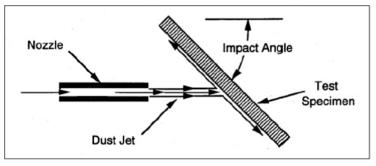
Plume/ Surface Interactions



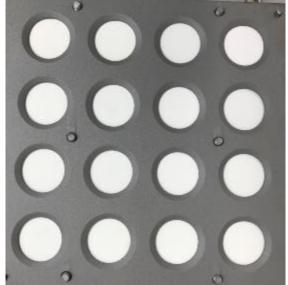
Sample configuration for paint, flex cable, and

M55J composite

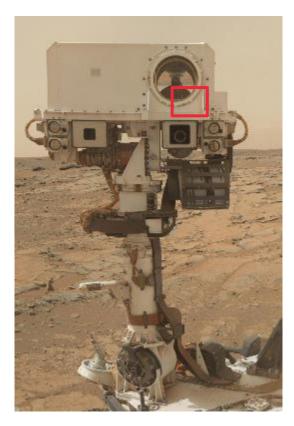




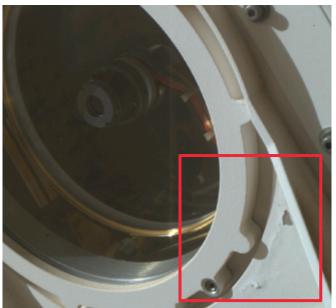




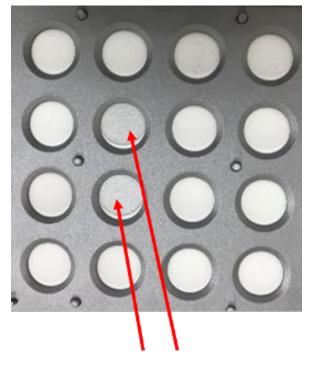
Failure of Aptek 2711 (MSL Paint)



Sol 177



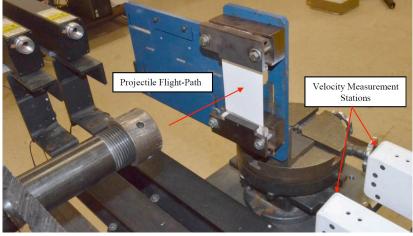
Sol 601

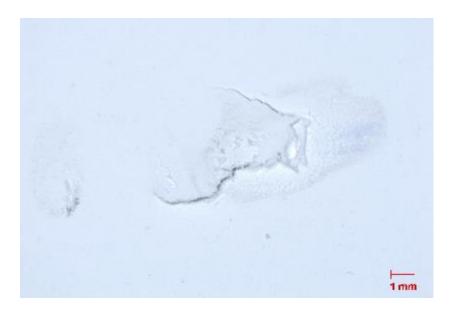


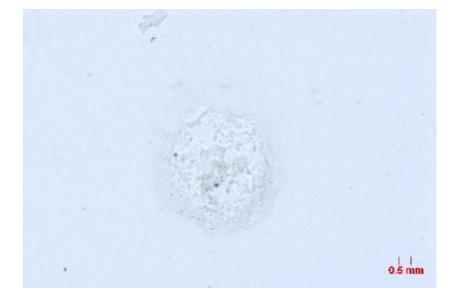
MSL Paint shows bare Al

Gas Gun Setup and Particle Impact







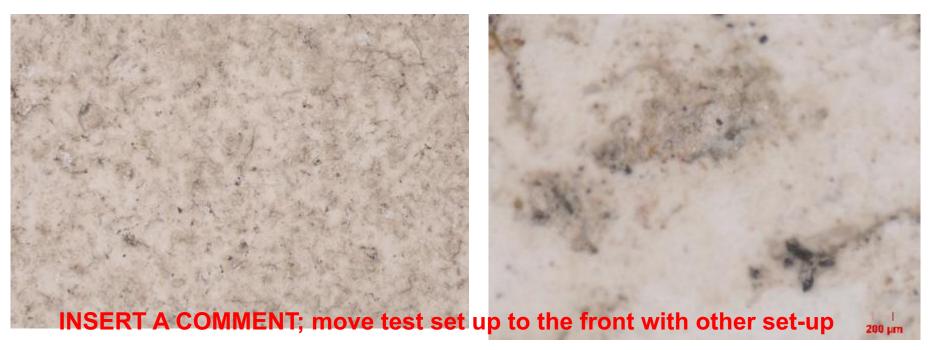


15.8 mm particle impact at 30°

15.8 mm particle impact at 90°

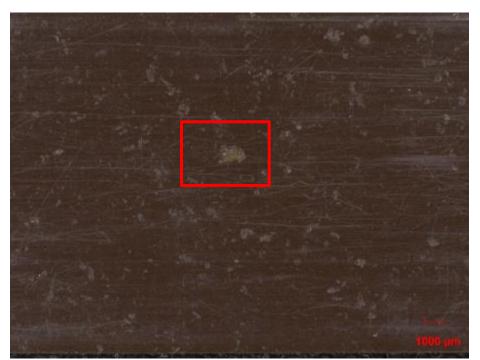
Gravelometer Erosion of S13

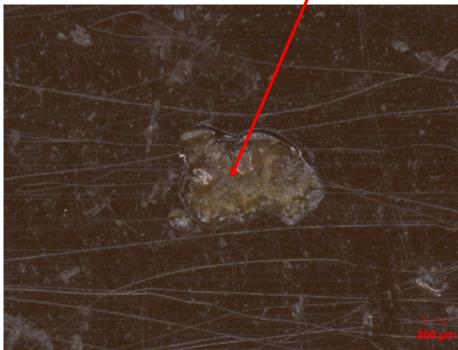




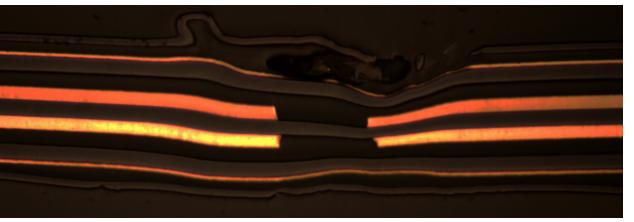
Flex Cable Gravelometer Erosion

Metal trace is not exposed

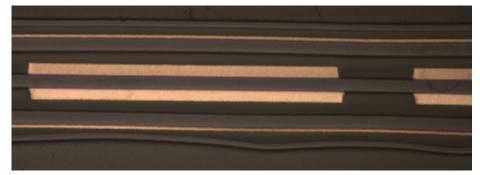




Cross-section shows no evidence of shorts or opens

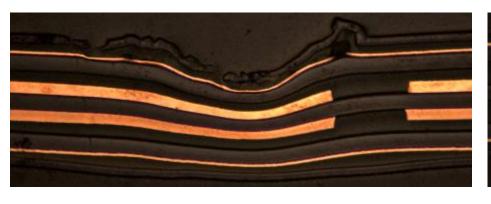


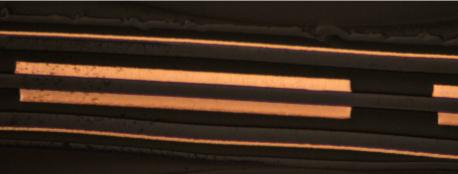
Flex Cable Cross-Sections





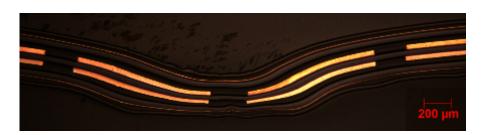




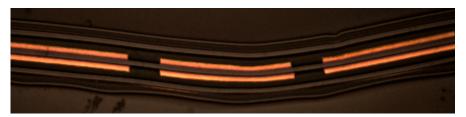


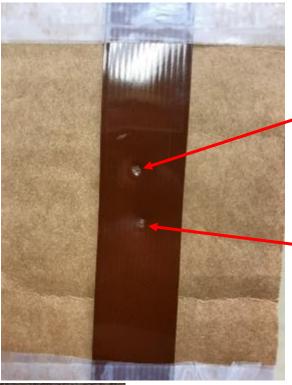
Flex Cable Gas Gun Particle Impact

Flex attached to Al



Flex attached to Cardboard





Impact with Al behind Flex Cable

Impact with cardboard behind the Flex Cable

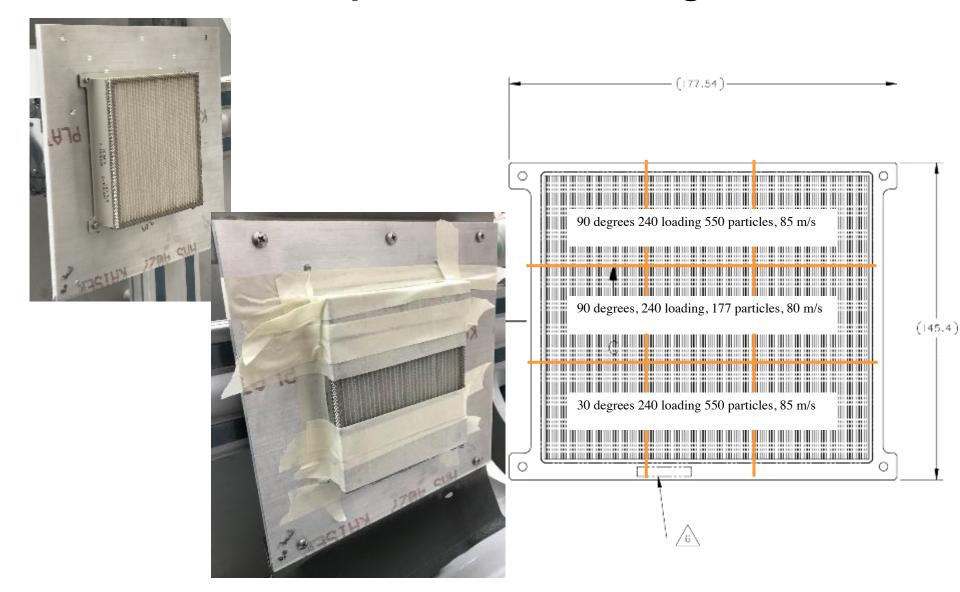


M55J Composite Sand Erosion



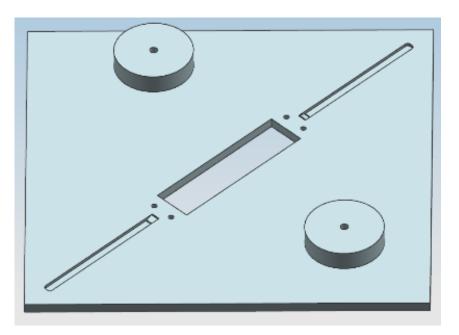


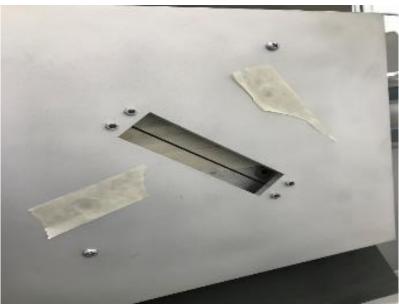
HEPA Filter Setup in the Erosion Rig



40

Fiber Optic Cable Setup in the Erosion Rig



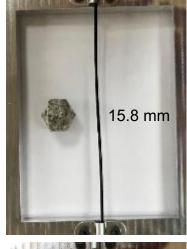




Fiber Optic Cable Gas Gun Setup



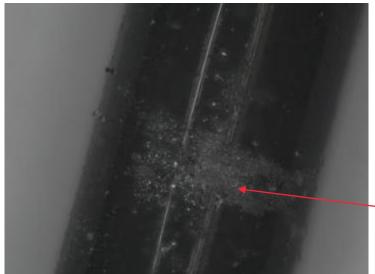




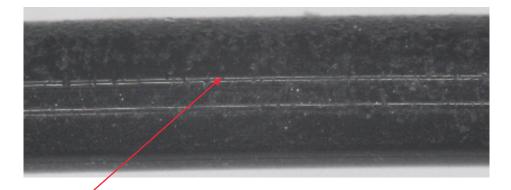


- Fiber Optic Cable deformed about 2" after impact from a 15.8 mm particle traveling at 20 m/s
- Maximum deformation according to manufacture is 2"
- Preliminary attenuation measurement show light dimmer than pre-test
- Post-Test attenuation measurement all pass but one

Fiber Optic Cable







Particles imbedded in the FOC jacket and scratches

Indentation in the FOC jacket

